

Satellite Laser Ranging and Rules of the Road for the International Laser Ranging Service

<http://ilrs.gsfc.nasa.gov/index.html>

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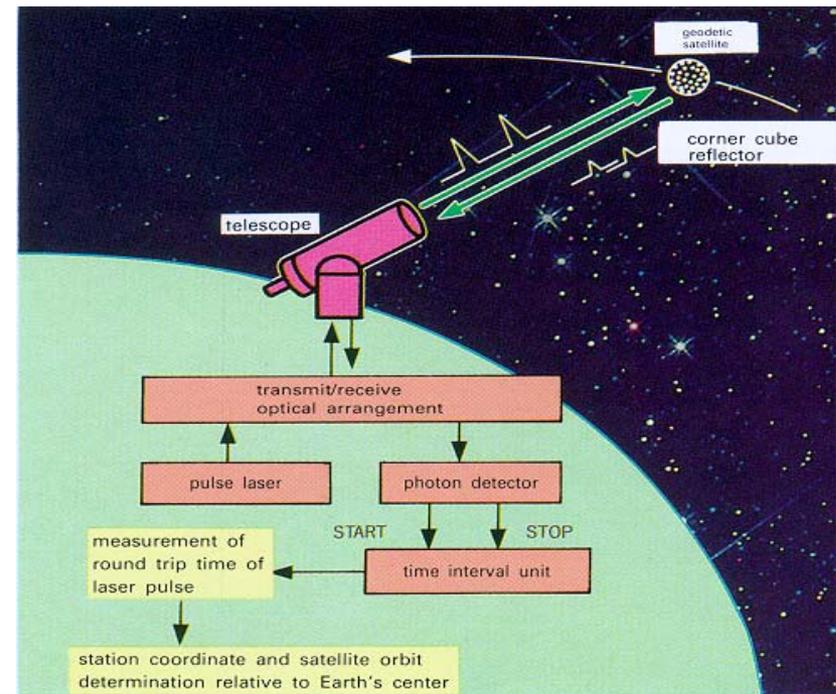
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Satellite Laser Ranging Technique

Precise range measurement between an SLR ground station and a retroreflector-equipped satellite using ultrashort laser pulses corrected for refraction, satellite center of mass, and the internal delay of the ranging machine.

- Simple range measurement
- Space segment is passive
- Simple refraction model
- Night / Day Operation
- Near real-time global data availability
- Satellite altitudes from 400 km to synchronous satellites, and the Moon
- Cm satellite Orbit Accuracy
- Able to see small changes by looking at long time series



- Unambiguous centimeter accuracy orbits
- Long-term stable time series

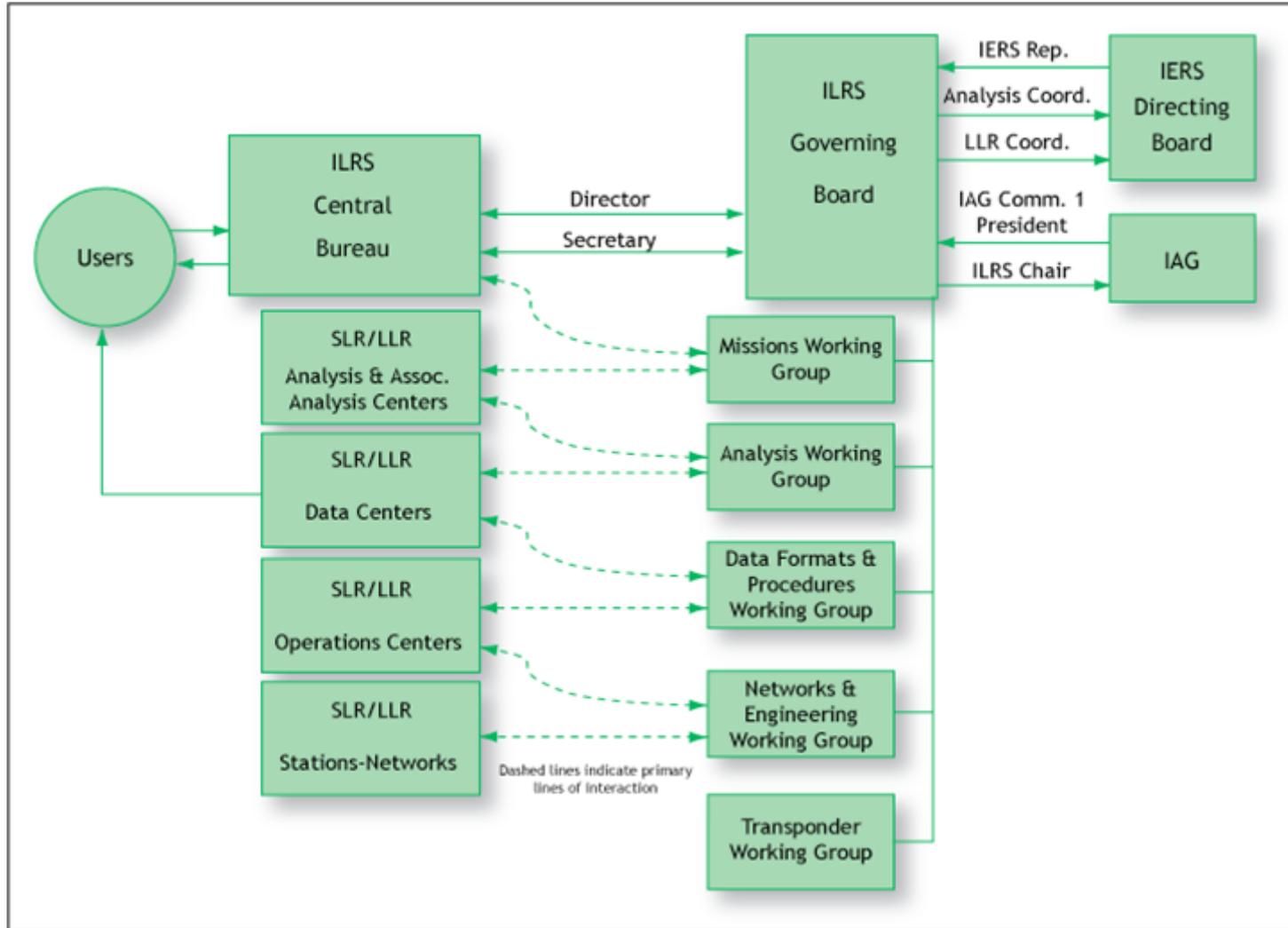
SLR Science and Applications

- Measurements
 - Φ Precision Orbit Determination (POD)
 - Φ Time History of Station Positions and Motions
- Products
 - Φ Terrestrial Reference Frame (Center of Mass and Scale)
 - Φ Plate Tectonics and Crustal Deformation
 - Φ Static and Time-varying Gravity Field
 - Φ Earth Orientation and Rotation (Polar Motion, length of day)
 - Φ Orbits and Calibration of Altimetry Missions (Oceans, Ice)
 - Φ Total Earth Mass Distribution
 - Φ Space Science - Tether Dynamics, etc.
 - Φ Relativity Measurements and Lunar Science
- More than 60 Space Missions Supported since 1970
- Four Missions Rescued in the Last Decade

International Laser Ranging Service (ILRS)

- Established in 1998 as a service under the International Association of Geodesy (IAG)
- ILRS collects, merges, analyzes, archives and distributes satellite and lunar laser ranging data to satisfy a variety of scientific, engineering, and operational needs and encourages the application of new technologies to enhance the quality, quantity, and cost effectiveness of its data products
- Components
 - ◻ Tracking Stations and Subnetworks
 - ◻ Operations Centers
 - ◻ Global and Regional Data Centers
 - ◻ Analysis and Associate Analysis Centers
 - ◻ Central Bureau
- ILRS produces standard products for the scientific and applications communities
- ILRS includes 75 agencies in 26 countries

ILRS Organization



Complex of Space Geodesy instruments

for development and maintenance of the reference frame



SLR/LLR



VLBI

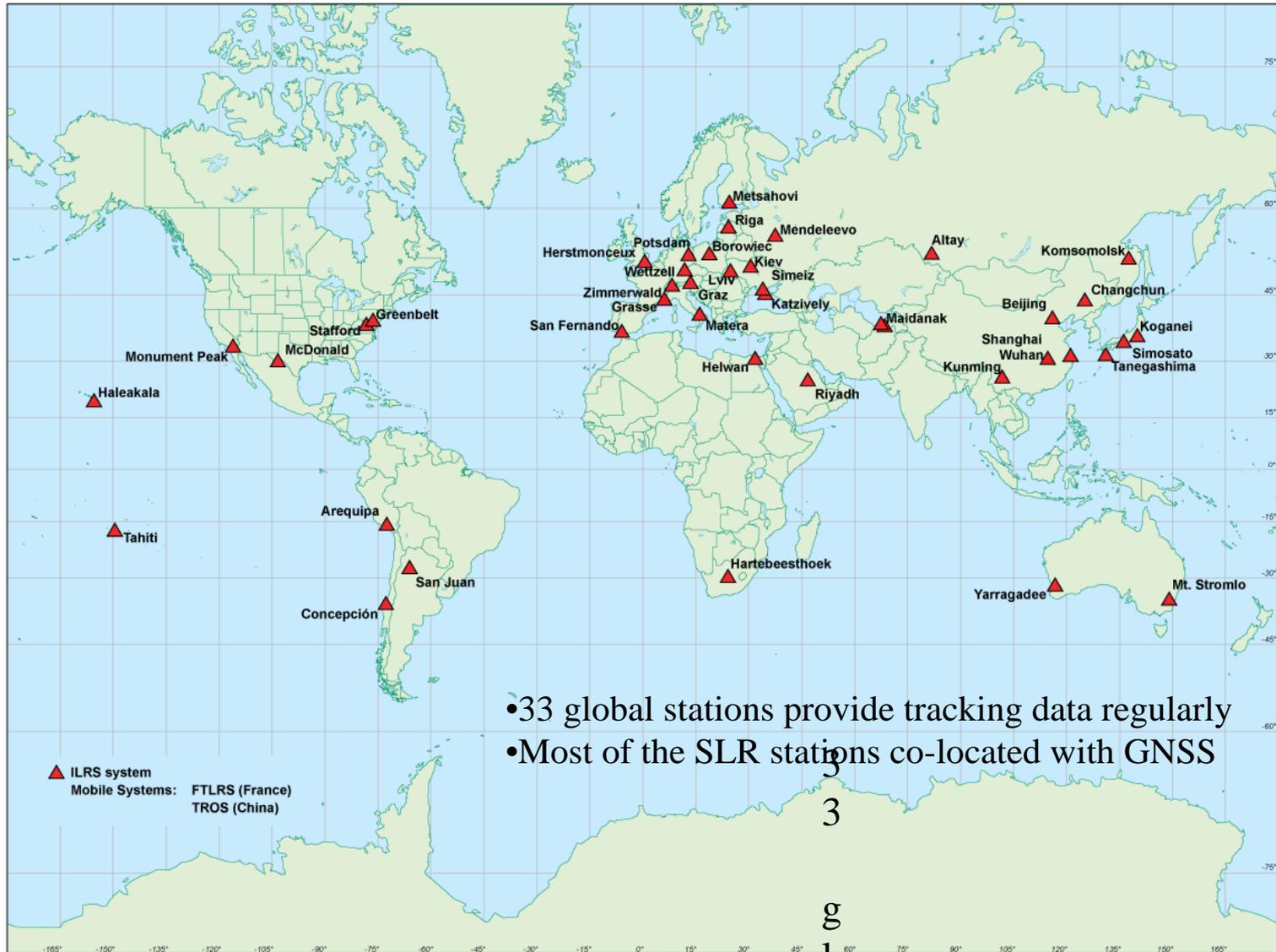


GPS



DORIS

ILRS Network



- 33 global stations provide tracking data regularly
- Most of the SLR stations co-located with GNSS

3

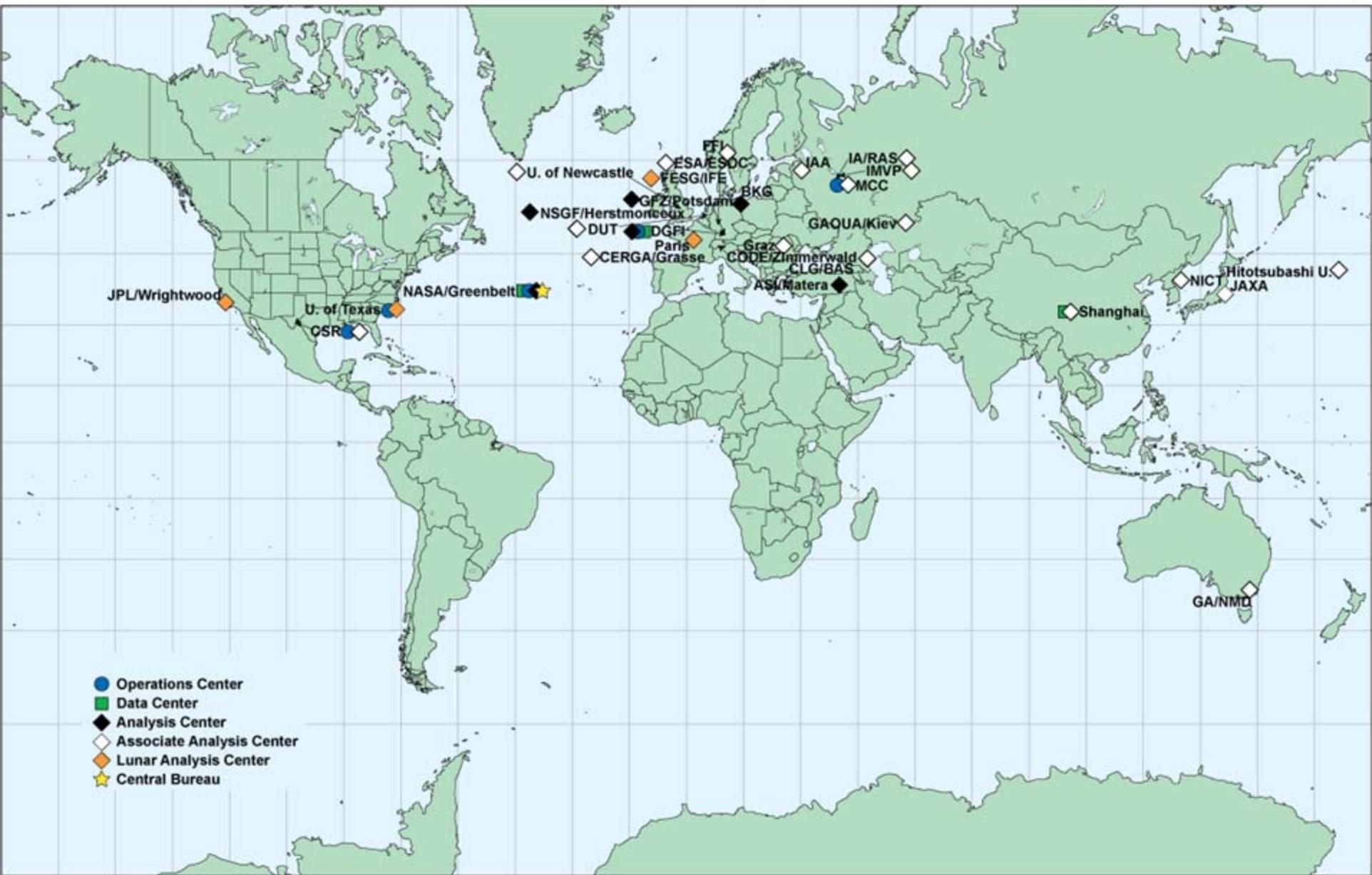
g

1

o

b

Other ILRS Components



Selected SLR Stations Around the World



NASA New Generation SLR System



NASA's Next Generation SLR (NGSLR), GGAO, Greenbelt, MD

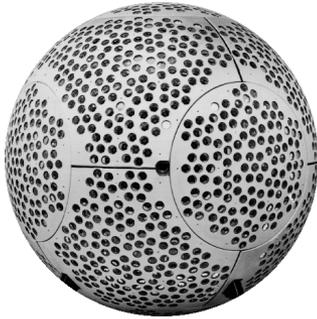
Technology Developments

- 2 kHz operation to increase data yield and improve interleaving
- Eye-safe operations and auto tracking
- Automation (unattended operation)
- Event timers with near-ps resolution
- Web-based restricted tracking to protect optically vulnerable satellites (ICESat, ALOS, etc.)
- Two wavelength experiments to test refraction models
- Experiments continue to demonstrate optical transponders for interplanetary ranging
 - ⊕ Transponder experiment to Messenger (24.3 million km) was a two-way demonstration that resulted in a range precision of less than 20 cm.
 - ⊕ Mars Global Surveyor MOLA experiment (over 80 million km link) was a one-way demonstration due to an inoperative laser at Mars.

Sample of SLR Satellite Constellation (Geodetic Satellites)



Etalon-I & -II



LAGEOS-1



LAGEOS-2



Ajisai



Starlette



Stella

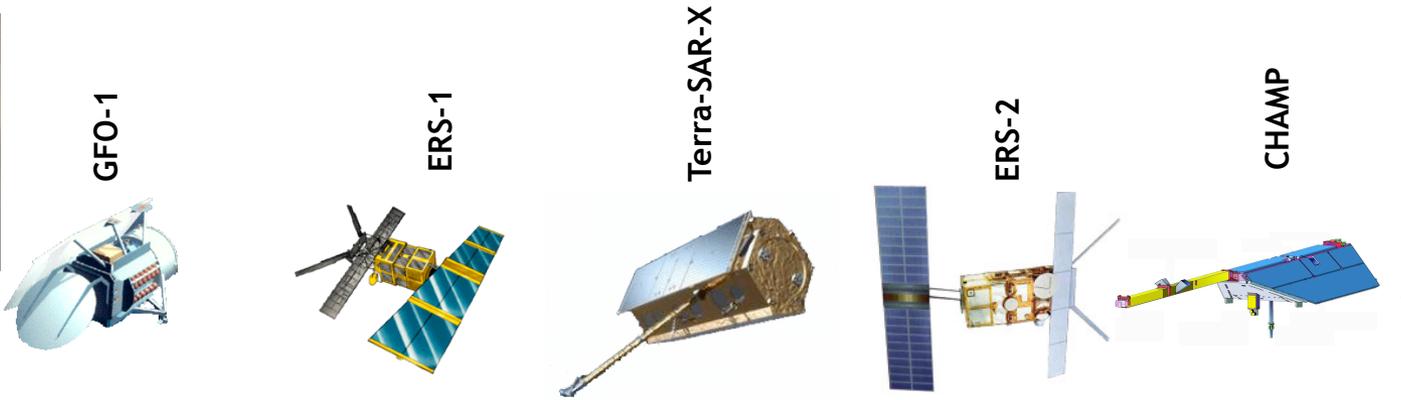
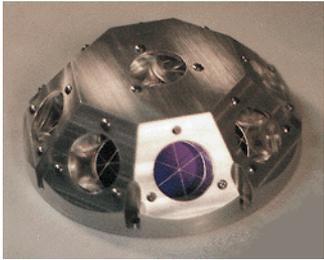


GFZ-1



| | | | | | | | |
|------------------|--------|--------|-------|-------|------|-------|-------|
| Inclination | 64.8° | 109.8° | 52.6° | 50° | 50° | 98.6° | 51.6° |
| Perigee ht. (km) | 19,120 | 5,860 | 5,620 | 1,490 | 810 | 800 | 396 |
| Diameter (cm) | 129.4 | 60 | 60 | 215 | 24 | 24 | 20 |
| Mass (kg) | 1415 | 407 | 405.4 | 685 | 47.3 | 47.3 | 20.6 |

Sample of SLR Satellite Constellation



| | | | | | |
|-------------------------|-------------|--------------|--------------|--------------|---------------|
| Inclination | 108° | 98.5° | 97.4° | 98.5° | 87.27° |
| Perigee ht. (km) | 800 | 780 | 514 | 785 | 474 |
| Mass (kg) | 300 | 2,400 | 1,230 | 2,516 | 400 |



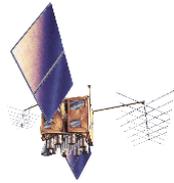
| | | | | | |
|-------------------------|---------------|--------------|-----------------|--------------|--------------|
| Inclination | 99.64° | 66° | 89° | 98.5° | 51.6° |
| Perigee ht. (km) | 1,012 | 1,336 | 450 | 796 | 250 |
| Mass (kg) | 2.477 | 500 | 432/sat. | 8,211 | 50 |

Sample of SLR Satellite Constellation (HEO)

GLONASS



GPS



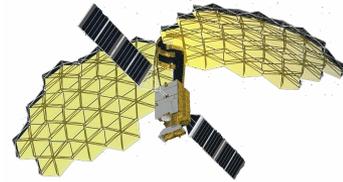
COMPASS



GIOVE



ETS-8



Inclination

65°

64.8°

55.5°

56°

0°

Perigee ht.
(km)

19,140

20,195

21,500

23,920

36,000

Mass (kg)

1,400

930

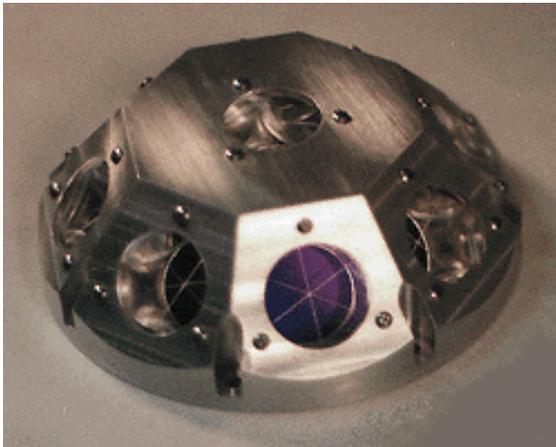
2,200

600

2,800



LAGEOS



Jason-2



Missions for 2009



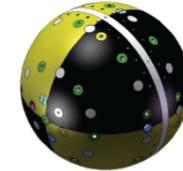
SOHLA
JAXA/Japan
January 2009



GOCE
ESA
March 2009



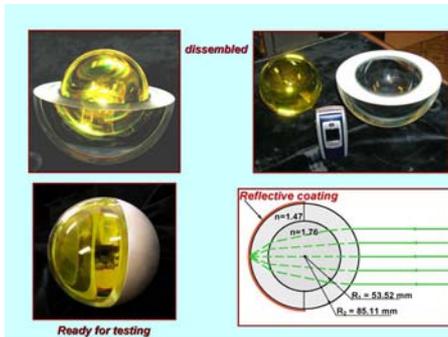
LRO
NASA/USA
April 2009



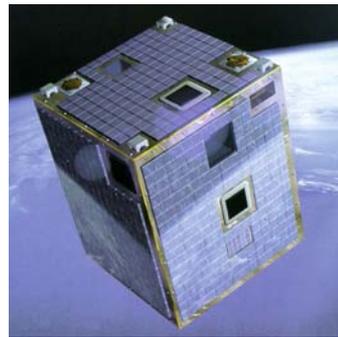
ANDE
NRL/USA
June 2009



STSAT-2
KASI/Korea
Mid-2009



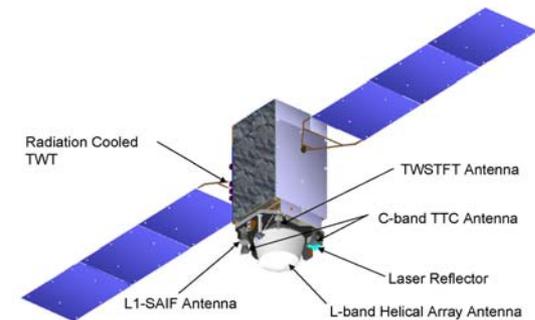
BLITS
IPIE/Russia
June 2009



PROBA-2
ESA
Mid-2009

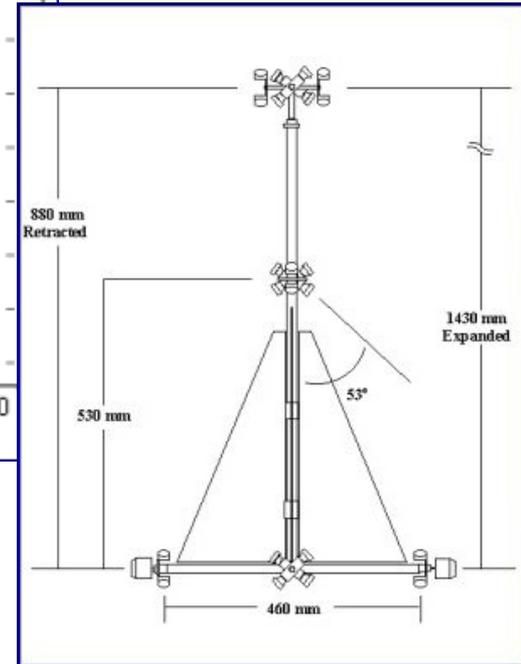
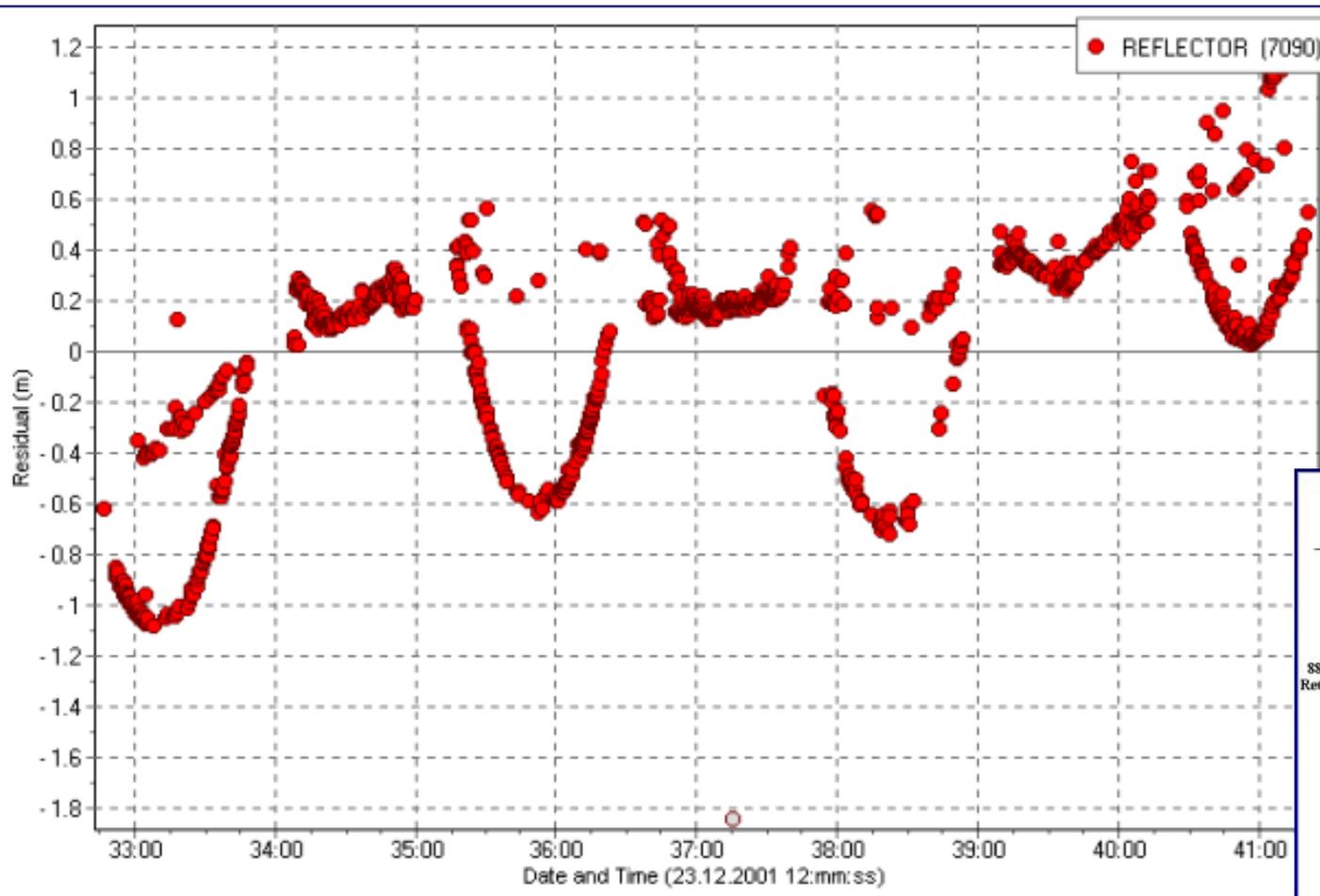


TanDEM-X
DLR, GFZ/Germany
2009

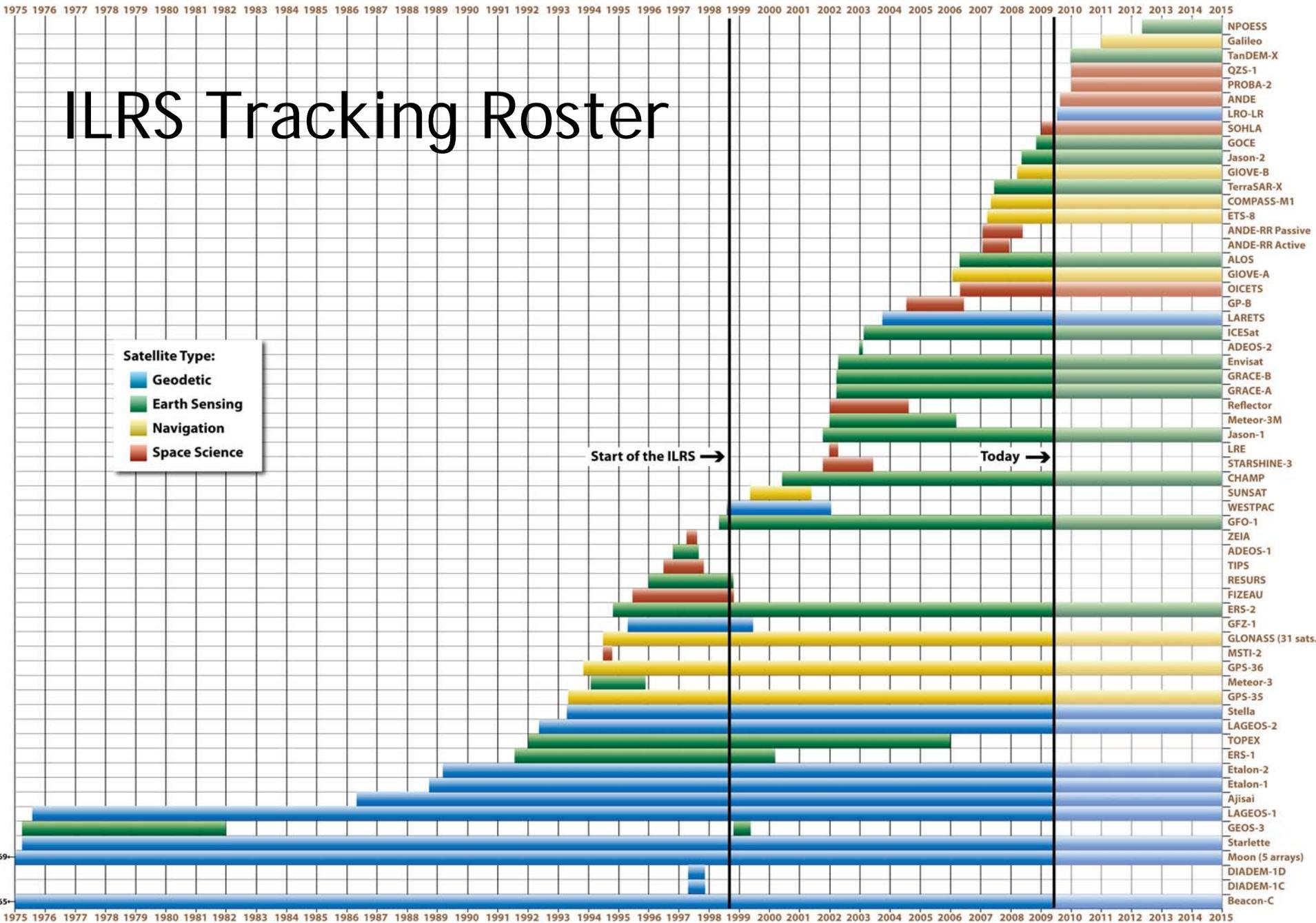


QZS-1
JAXA/Japan
2009

Reflector Satellite



ILRS Tracking Roster



Mission Requirements

- Submit Mission Support Request Form;
- Approval from the Missions Working Group and the Governing Board
 - Science and technology available to the community
 - Tracking requirements and practicality
- Regular updates through launch and insertion
- Predictions in ILRS format
- In orbit updates
- Periodic reports on mission status and progress, data requirements, and data fulfillment

Station Requirements

- Submit Site and System Information Form (Station characteristics)
- Approved by the Governing Board
- Format compatibility
- Inform the ILRS of programmed station downtime and station modifications and configuration changes
- Proper site survey
- Requirements for Operational Status (Data volume, accuracy, latency, etc.)
- Participation in ILRS activities - Biannual Report, Workshops, etc.

Aircraft Safety

- Radars
- Link to Air Traffic Control
- Visual spotters
- Optical detection systems
- Eye-safe laser systems
- Low energy - high repetition rate laser

ILRS Restricted Tracking

ILRS authorization to track ILRS-approved satellites is constituted and governed by an approved Mission Support Request Form;

All SLR stations within the International Laser Ranging Service agree to adhere to any applicable ILRS Restricted Tracking Procedures including:

- station by station authorization;
- time and viewing angle constraints;
- energy/power constraints;
- go/no-go switch.

Need for SLR measurements on the GNSS Constellations

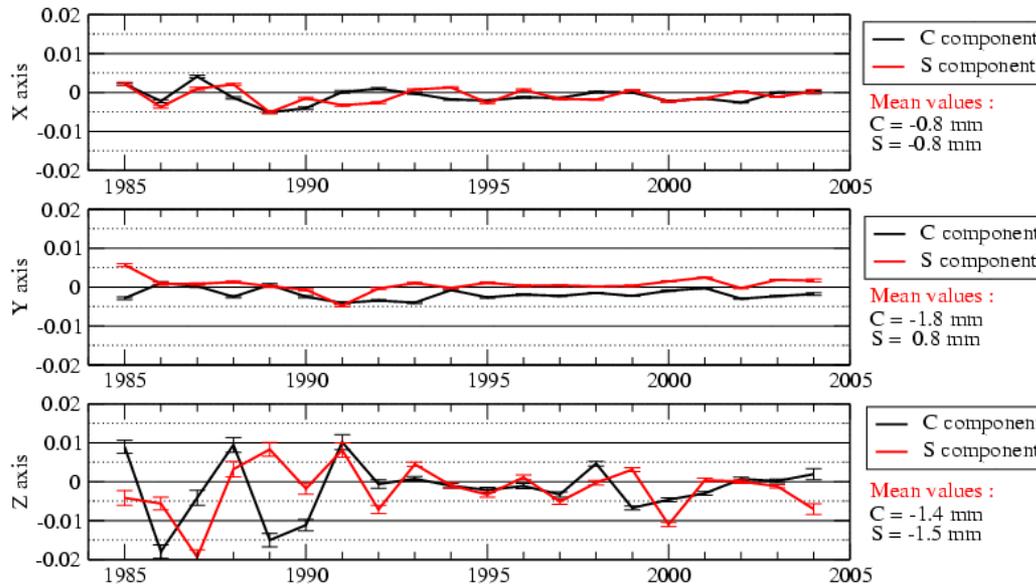
- Geoscience
 - ⊕ Improve the Terrestrial Reference Frame (colocation in space)
 - Basis on which we measure global change over space, time, and evolving technology
 - Relies on colocation measurements with different technologies -GNSS, VLBI, SLR, DORIS, ----
 - Most stringent requirements - ocean surface, ice budget
 - ⊕ Improve LEO POD
 - Altimeter satellites
- GNSS World
 - ⊕ Provide independent Quality Assurance: - The GNSS orbit accuracy cannot be directly validated from the GNSS data itself;
 - ⊕ Assure interoperability amongst GPS, GLONASS, Galileo, COMPASS --
 - ⊕ Insure realization of WGS84 reference frame is consistent with ITRF
 - ⊕ SLR is NOT required for use in routine / operational RF derived orbit and clock products

Current SLR Ranging to GNSS Satellites

- Operations include 8 GNSS satellites
(GPS 35 and 36; GLONASS 102, 109 and 115; GIOVE - A and - B; and *COMPASS*)
- Satellite priorities set according to satellite altitude;
- Track 5 minute segments at various points along the pass;
- Data transmitted after each pass;
- The data is available on the website within an hour or two;
- Plenty of spare SLR tracking capacity

Geocenter Motion

Geocentre motion, annual terms
(m)



Mean annual terms amount to :

1.2 mm in X, with a minimum in February

2.0 mm in Y, with a minimum in December

1.8 mm in Z, with a minimum in February

} corresponding to a winter loading centred on Siberia

- mm-level Geodesy requires understanding of the reference frame and its distortions to acute levels of precision.
- Shown here is the change in the origin of the crust-fixed frame w.r.t. the center of mass due to non tidal mass transport in the atmospheric and hydrospheric systems.

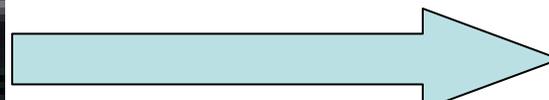
One-Way Earth-to-Mars Transponder Experiment

(September 2005)



GSFC 1.2 Meter Telescope

80 Million Km!



~500 laser pulses
observed at Mars!



Mars Orbiter Laser Altimeter (MOLA)

Ground Station

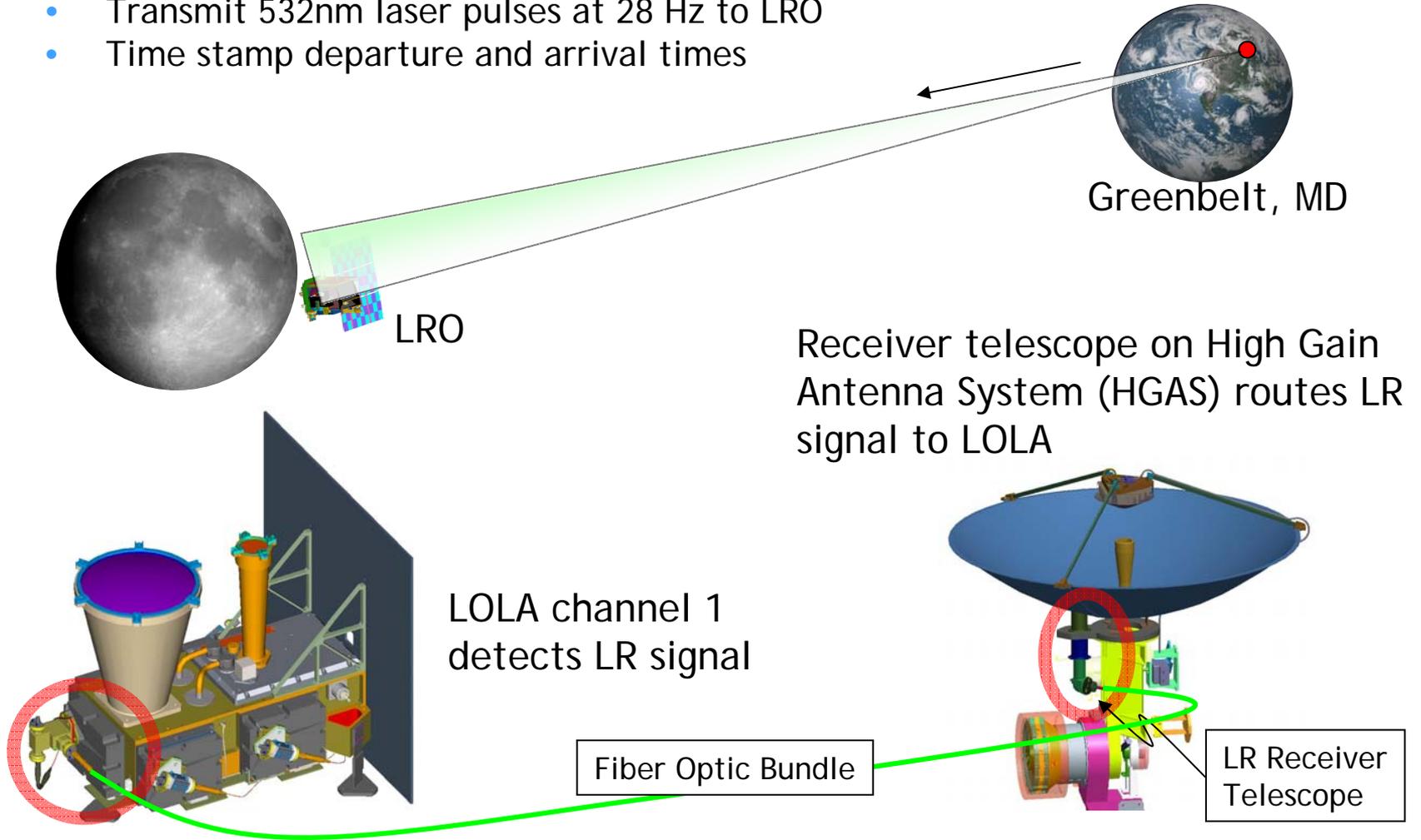
Xiaoli Sun Jan McGarry
Tom Zagwodzki John Degnan

Science/Analysis/Spacecraft

David Smith Maria Zuber
Greg Neumann Jim Abshire

LRO Laser Ranging

- Transmit 532nm laser pulses at 28 Hz to LRO
- Time stamp departure and arrival times



Some Transponder Applications

- **Solar System Science**
 - ⊕ Solar Physics: gravity field, internal mass distribution and rotation
 - ⊕ Lunar ephemerides and librations
 - ⊕ Planetary ephemerides
 - ⊕ Mass distribution within the asteroid belt
- **General Relativity**
 - ⊕ Tests of relativity and constraints on its metrics Precession of Mercury's perihelion
 - Constraints on the magnitude of \dot{G} (1×10^{-12} from LLR)
 - Gravitational and velocity effects on spacecraft clocks
 - Shapiro Time Delay
- **Lunar and Planetary Mission Operations**
 - ⊕ Spacecraft ranging
 - ⊕ Calibration/validation/backup for DSN microwave tracking
 - ⊕ Subnanosecond transfer of GPS time to interplanetary spacecraft for improved synchronization of Earth/spacecraft operations
 - ⊕ Independent self-locking beacon for collocated laser communications systems (e.g., NASA's Mars Laser Communications Demonstration)

We invite you to visit our website @

<http://ilrs.gsfc.nasa.gov/index.html>